

1. Perform simulation of an airfoil.

```
cp -r $FOAM_TUTORIALS/incompressible/simpleFoam/airFoil2D $FOAM_RUN
cd $FOAM_RUN/airFoil2D
# Study files with ls, less, find, etc.
checkMesh > log.checkMesh.1
renumberMesh -overwrite >log.renumberMesh
checkMesh > log.checkMesh.2
# Study the log files!
meld log.checkMesh*
# Study the difference!
simpleFoam > log.simpleFoam.1
# Study new files with ls, less, find, etc.
# Study the solver logfile!
tail log.simpleFoam.1
# Note that the solution is converged.
```



2. Create a changeDictionary file to change flow velocity.

```
head -n 17 system/controlDict > system/changeDictionaryDict
sed -i 's/controlDict/changeDictionaryDict/' system/changeDictionaryDict
echo "dictionaryReplacement
```

```
U
       boundaryField
            inlet
                freestreamValue uniform (15 -3 0);
            outlet
                freestreamValue uniform (15 -3 0);
}" >> system/changeDictionaryDict
```



3. Change boundary condition and continue simulation.

cp 350/U 350/U.org changeDictionary meld 350/U\*# Find all differences! simpleFoam > log.simpleFoam.2 # Study the solver logfile! tail log.simpleFoam.2 # Note that the solution is NOT converged. cp system/controlDict system/controlDict.org sed -i 's/500/1000/' system/controlDict meld system/controlDict\* simpleFoam > log.simpleFoam.3 # Study the solver logfile! tail log.simpleFoam.3 # Note that the solution is now converged. paraFoam



4. Visualize streamlines for the two converged solutions!

- Stream tracer, Line source, Y axis
- Time = 350 s and 783 s

5. Plot residuals and number of iterations using gnuplot!

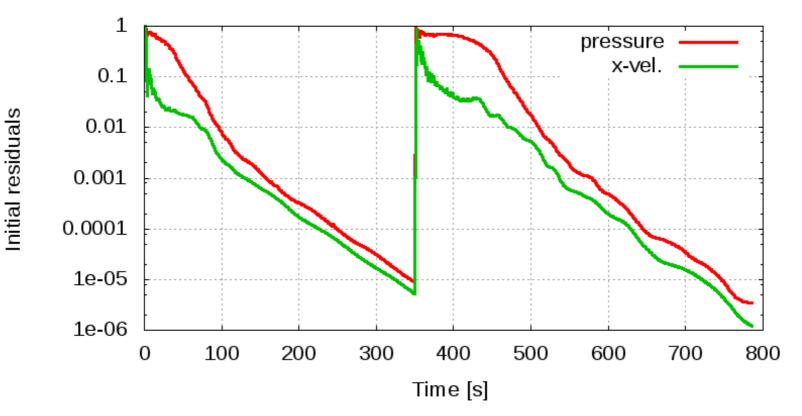
```
cat log.simpleFoam.* > log.simpleFoam
foamLog log.simpleFoam # Study files in logs directory!
gnuplot
```

```
plot 'logs/pIters_0' w l, 'logs/UxIters_0' w l, 'logs/UyIters_0' w l
set grid
plot 'logs/p_0' w l lw 3 t 'pressure', 'logs/Ux_0' w l lw 3 t 'x-vel.'
set term png size 800,450 font Arial 16
set out 'residuals.png'
set logscale y
set title 'Convergence of solution'
set xlabel 'Time [s]'
set ylabel 'Initial residuals'
replot
exit
```



6. Open the png file and study the plot!

#### display residuals.png



Convergence of solution



6. Perform a simplified motorbike simulation and visualize results.

```
cp -r $FOAM_TUTORIALS/incompressible/simpleFoam/motorBike $FOAM_RUN
cd $FOAM RUN/motorBike
sed -i 's/20 8 8/5 2 2/' constant/polyMesh/blockMeshDict
cp $FOAM_TUTORIALS/resources/geometry/motor* constant/triSurface/
surfaceFeatureExtract > log.surfaceFeatureExtract
blockMesh > log.blockMesh
snappyHexMesh -overwrite > log.snappyHexMesh
cp -r 0.org/* 0/
potentialFoam -noFunctionObjects -writep > log.potentialFoam
simpleFoam >log.simpleFoam
vorticity
foamCalc mag U
foamCalc components U
paraFoam
# Stream tracer, Line source, e.g. z axis
# Slice, e.g. y plane
# Study and learn features of ParaView, e.g. Contour, Clip, Opacity!
# Enable all Volume Fields, e.g. vorticity!
```



# Assignments

- 1. Some utilities need dictionary files, e.g. *autoRefineMesh*. Where can you find an example dictionary file in OpenFOAM 2.3.1 installation?
- 2. What are the dimensions of field p in the *airFoil2D* case?
- 3. What is the average of velocity magnitude over the inlet patch in *airFoil2D*?
- 4. What is the integrated volume flow over the inlet patch in *airFoil2D*?
- 5. What are the dimensions of field B in the mhdFoam tutorial called hartmann?
- 6. Is the lift coefficient positive or negative for (15 -3 0) freestream velocity in *airFoil2D* case?



### Homework

- 1. Read OpenFOAM UserGuide chapter 3.5-3.7.
- 2. Perform *motorBike* simulation again at home. Try to create better visualization in ParaView, save colourful images and send images to teachers by e-mail.
- 3. Use sample utility to post-process results of *motorBike* simulation. You can read OpenFOAM UserGuide chapter 6.5 if you need help. Use gnuplot to create png images of profiles.