### **Boundary conditions**

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### What do we mean by boundary conditions?

$$\frac{\partial}{\partial t} \int\limits_{V} \rho \, \phi \, dV + \oint\limits_{A} \rho \, \phi \, \overline{v} \cdot d\overline{A} = \oint\limits_{A} (\overline{S}_A) \underbrace{\Gamma \, \nabla \phi} \cdot d\overline{A} + \int\limits_{V} S_V \ dV$$

The generic conservation equation in differential form:

$$\frac{\partial \rho \phi}{\partial t} + \nabla \cdot (\rho \phi \, \overline{v}) = \nabla \cdot \overline{S}_{A} + \nabla \cdot (\Gamma \, \nabla \phi) + S_{v}$$

3. Mixed BC: linear combination of  $\phi$  and it's normal derivative is given.

It is second order in space due to this term

Three types of boundary are possible for such PDEs: 1. BC of first kind: value of  $\phi$  is given at the boundary;

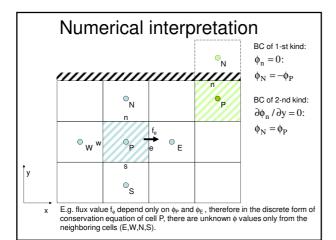
 $\frac{\partial \phi}{\partial x} = \nabla \phi \Big|_{\mathbf{n}}$ 

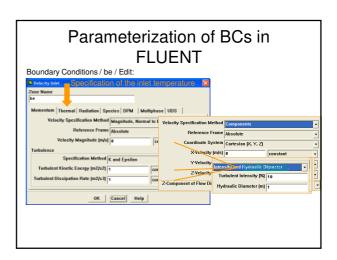
2. BC of second kind: normal derivative of φ is given;

Boundary conditions for each conservation equations cannot be independently

defined (E.g. we cannot use BC of 1-st kind for every velocity components along with BC of 1-st kind for pressure.)

Therefore FLUENT provides only boundary condition "packages" of well defined physical meaning.





### Inlet and outlet BC packages $_{\iota: \text{incompressible}}$ $_{\iota: \text{compressible}}$ Inflow (or outflow, if negative) of given velocity. BC of second kind for the pressure. First kind BC for every other field variable. Velocity-inlet Inflow at given mass-flow-rate or mass flux (pv) profile. It is a BC of Mass-flow-inlet ¿+c second kind for the pressure. Inflow with a given pressure profile. Flow direction must be specified. BC of 2nd kind for the velocity magnitude. BC of 1st kind for every othe scalar quantity. Outflow with a given static pressure profile. BC of 2nd kind for the Pressure-outlet ¿+¢ velocity and other field variables. Target mass-flow-rate can be spec Outflow with a given share of flow rate. BC of 2nd kind for every quantity. Non-reflective. Only outflow is allowed! In- or outflow with given far-field characteristics. Flow direction and Mach number can be specified. Non-reflective. Pressure-far-field Inlet-vent $\ell+c$ Pressure-inlet + $\zeta(v)$ loss coeff. E.g. an intake with a filter of grid. $_{\ensuremath{\ell^+e}}$ Pressure-inlet + $\Delta p(v)$ pressure rise. (Characteristic curve.) Outlet-vent Exhaust-fan e+e Pressure-outlet + $\Delta p(v)$ pressure rise

# Important notes

- Outflow cannot be used in the presence of Pressure Inlet or Pressure Outlet;
- Back flow is not allowed through an Outflow (due to immediate convergence problems);
  Velocity Inlet provides unphysical results in compressible flow simulations (Mass Flow Inlet need to be used in these cases);
- Pressure Inlet is automatically changed to Pressure Outlet when back flow occurs (and the Pressure Outlet does similarly);
- In the case of supersonic flow, only upstream values are used at the boundary surfaces;
- There are three ways of branching the flow:

  Outflow (with Flow Rate Weighting)
  - Multiple Pressure Outlets
  - Velocity Inlet with negative velocity (mathematically incorrect by works if proper care is taken)

# Symmetry Face normal velocity is 0. BC of 2<sup>nd</sup> kind for every other field variable. Wall BC of 1<sup>st</sup> kind for the velocity and 2<sup>nd</sup> kind for the pressure. Many options for thermal BCs. E.g. water surface can be modeled as a frictionless wall. Axis Axis of an axisymmetric 2D model. Always along the x axis. v=0, w=0, BC of 2<sup>nd</sup> kind for every other field variable. Periodic Flow quantities are matched in every point of the periodic surface pair. Requires identical mesh on both surface. Translational (eg. tube bundle) or rotational (eg. turbine blade) Pressure gradient or target mass-flowrate can be specified. Interface Connects separate grids. It does not require matching nodes. Grids can slide on each other in every time step. There is an option for periodic BC.

# Modeling fluid machinery

- 1. Actuation disk fan
- 2. Rotating frame frozen rotor
- 3. Bidirectional averaging mixing plane
- 4. Rotating zone sliding mesh

