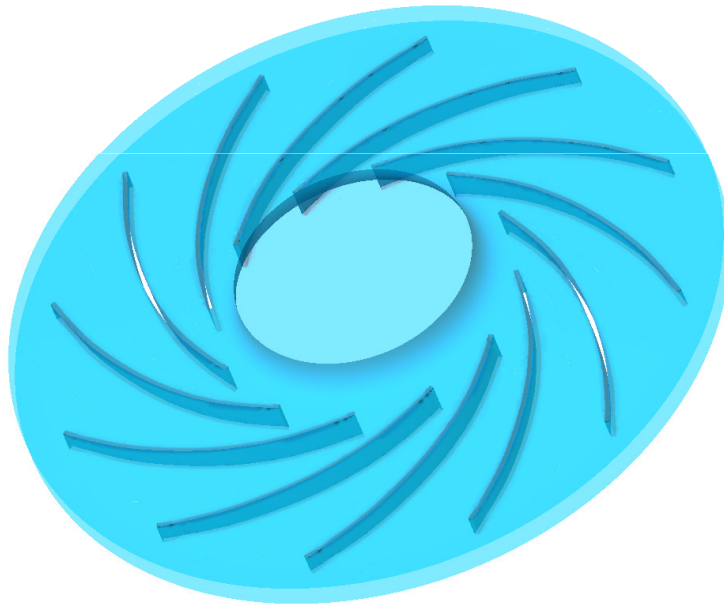


# 5. TURBOBLOWERS AND TURBOCOMPRESSORS

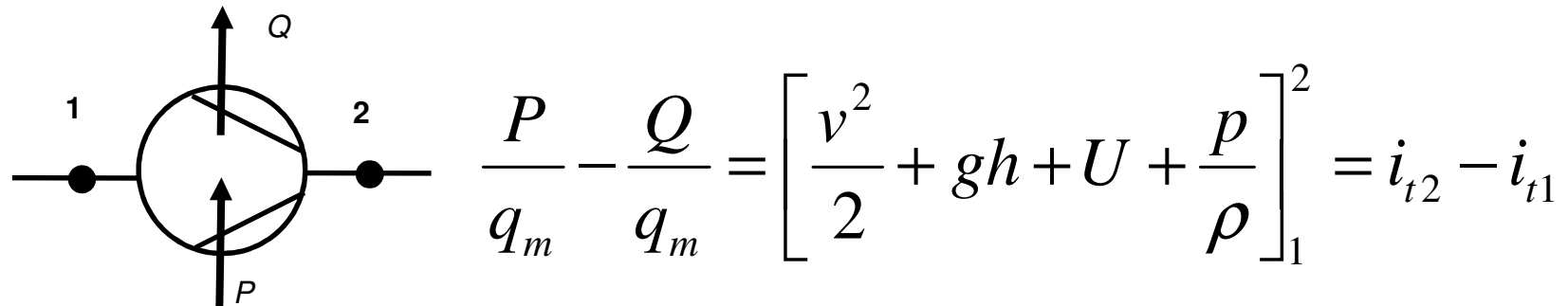
## 5.1. Blowers

$Q$ [l/s]	$n$ [1/min]	$\Delta p_t$ [Pa]	$\psi t$
32,5	32500	24386	1,16



CFX

## 5.1.2. Work process



$$\frac{P}{q_m} - \frac{Q}{q_m} = \left[ \frac{v^2}{2} + gh + U + \frac{p}{\rho} \right]_1^2 = i_{t2} - i_{t1}$$

$$P = q_m \left[ \frac{v^2}{2} + U + \frac{p}{\rho} \right]_1^2 \quad U = c_v T \quad \frac{p}{\rho} = RT = (c_p - c_v) T$$

$$P = q_m c_p \left[ \frac{v^2}{2c_p} + T \right]_1^2 \quad \frac{v^2}{2c_p} + T = T_{din} + T = T_t$$

$$P = q_m c_p (T_{t2} - T_{t1})$$

$$P = q_m c_p (T_2 - T_1)$$

### 5.1.3. Energetic aspects

$$\eta_e = \frac{c_p dT - dq}{c_p dT}$$

$$RT = p/\rho$$

$$dq = c_v dT + p d\left(\frac{1}{\rho}\right)$$

$$R dT = p d\left(\frac{1}{\rho}\right) + dp \left(\frac{1}{\rho}\right)$$

$$dq = c_v dT + R dT - dp \left(\frac{1}{\rho}\right) = c_p dT - dp \left(\frac{1}{\rho}\right)$$

$$\eta_e = \frac{dp \frac{1}{\rho}}{c_p dT}$$

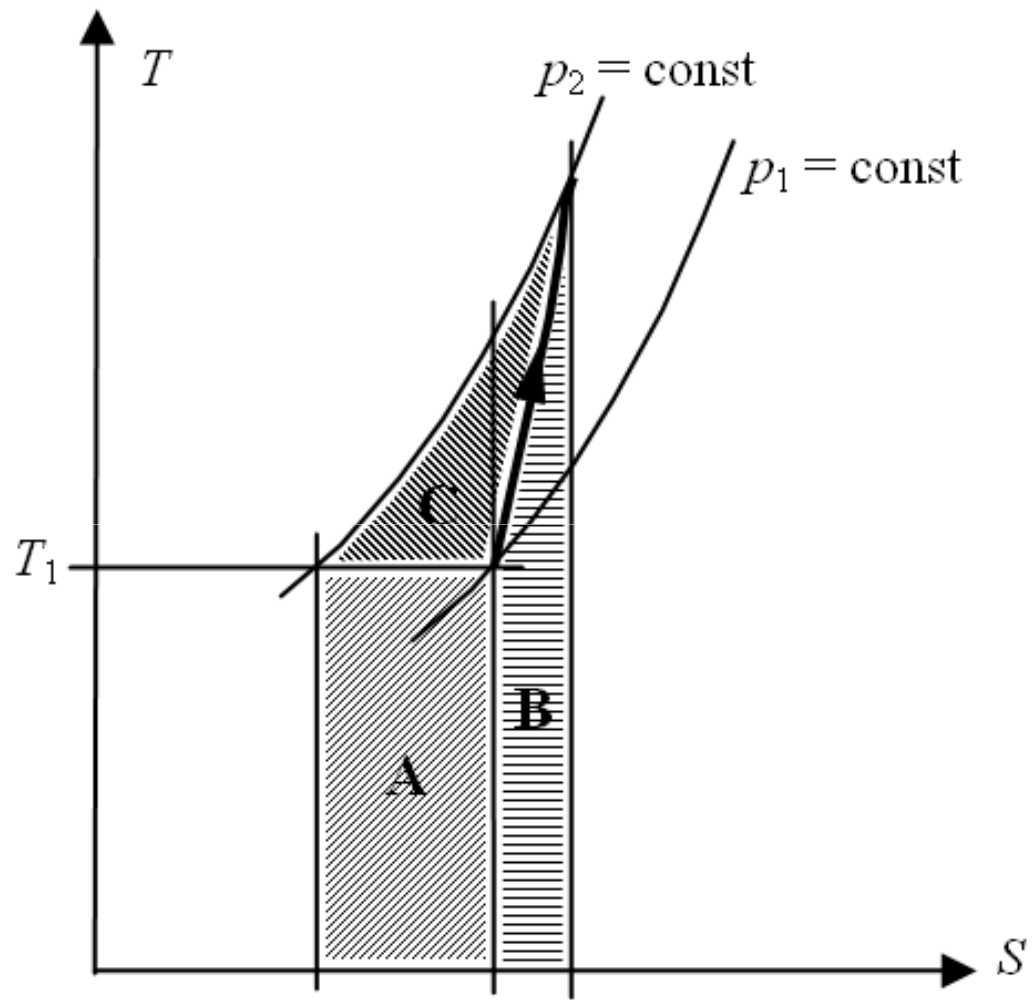
$$\frac{1}{\rho} = \frac{RT}{p}$$

$$\eta_e = \frac{dp RT}{c_p dT p}$$

$$\frac{dT}{T} = \frac{R}{c_p \eta_e} \frac{dp}{p}$$

$$\eta_p = \frac{\kappa - 1}{\kappa} \frac{\ln \frac{p_2}{p_1}}{\ln \frac{T_2}{T_1}}$$

$$\frac{T_2}{T_1} = \left( \frac{p_2}{p_1} \right)^{\frac{\kappa - 1}{\kappa \eta_p}}$$



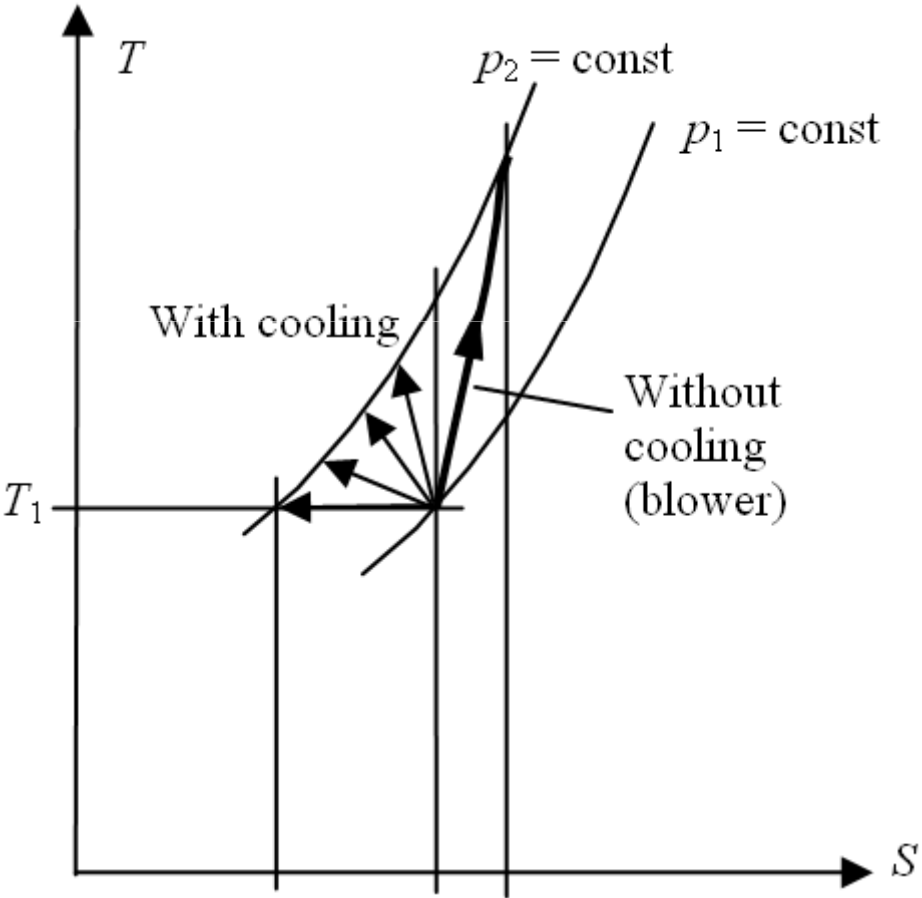
$$\eta_p = \frac{A + C}{A + B + C}$$

## Isothermal power factor:

$$\lambda = \frac{H_{isoth}}{\sum P/q_m}$$

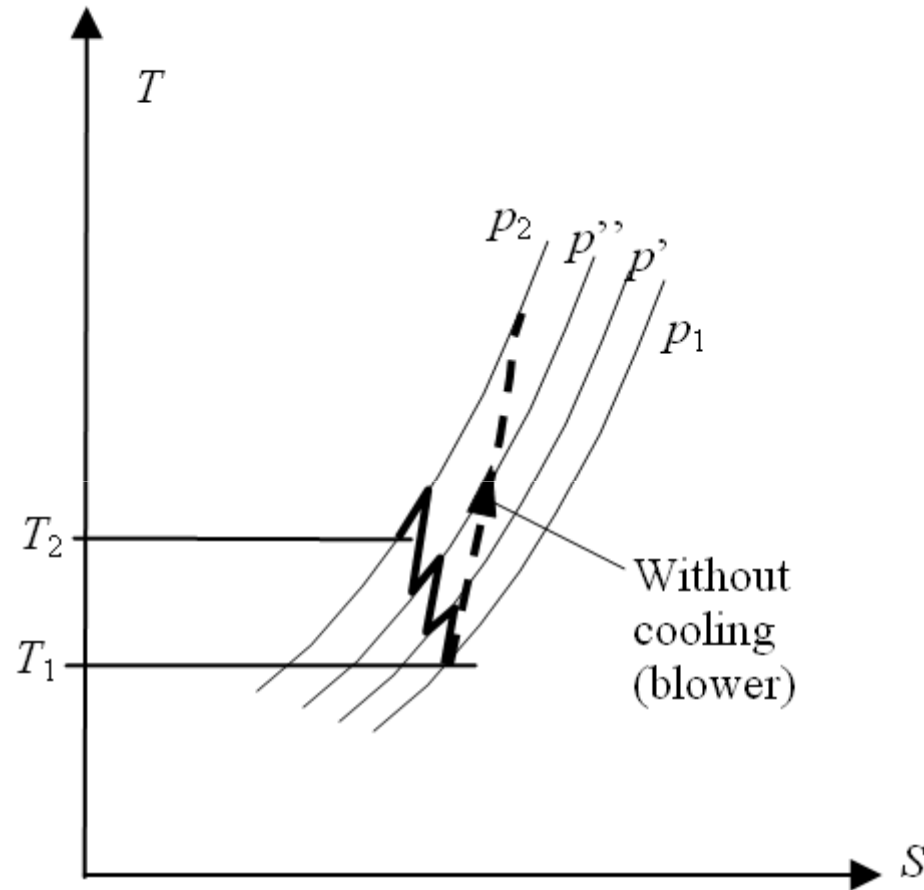
$$H_{isoth} = \int_1^2 \frac{1}{\rho} dp = RT_1 \int_1^2 \frac{1}{p} dp = RT_1 \ln \frac{p_2}{p_1}$$

# 5.2. Compressors



# Temperature coefficient:

$$\chi = \frac{\Delta T}{u^2 / 2c_p}$$



**2.**

**Blower:** pressure ratio  $p_2/p_1 = 1.6$

Inlet air:  $p_1 = 10^5 \text{ Pa}$ ,  $t_1 = 20 \text{ °C}$

Mass flow rate:  $q_m = 1 \text{ kg/s}$

Outlet temperature:  $t_2 = 70 \text{ °C}$

Air characteristics:  $c_p = 1000 \text{ J/(kg K)}$ ,  $\kappa = 1,40$ ,  $R = 287 \text{ J/(kg K)}$

A/ Polytropic efficiency:  $\eta_p = \frac{\kappa - 1}{\kappa} \ln \frac{p_2}{p_1} / \ln \frac{T_2}{T_1}$

B/ Outlet temperature for isentropic compression:  $T_{2IS} = T_1 \left( \frac{p_2}{p_1} \right)^{\frac{\kappa - 1}{\kappa}}$

(same pressure ratio)

C/ Shaft power input:  $P = q_m c_p (T_2 - T_1)$

D/ One stage: pressure ratio not more than  $\varepsilon = 1.1$

Necessary number of stages:  $N = ?$

$$p_2/p_1 = \varepsilon^N \quad \lg(p_2/p_1) = N \cdot \lg \varepsilon$$



3.

**Compressor:** pressure ratio  $p_2/p_1 = 4.0$

Inlet air:  $p_1 = 10^5 \text{ Pa}$ ,  $t_1 = 20 \text{ °C}$

Mass flow rate:  $q_m = 10 \text{ kg/s}$

Outlet temperature:  $t_2 = 80 \text{ °C}$

Isothermal power factor:  $\lambda = 0.68$

Air characteristics:  $c_p = 1000 \text{ J/(kg K)}$ ,  $\kappa = 1,40$ ,  $R = 287 \text{ J/(kg K)}$

A/ Outlet temperature for isentropic compression:  
(same pressure ratio)

$$T_{2IS} = T_1 \left( \frac{p_2}{p_1} \right)^{\frac{\kappa-1}{\kappa}}$$

B/ Useful isothermal specific work:  $H_{isoth} = RT_1 \ln \frac{p_2}{p_1}$

C/ Overall power consumption:  $\sum P = \frac{H_{isoth} q_m}{\lambda}$

D/ One stage: pressure ratio not more than  $\varepsilon = 1.2$

Necessary number of stages:  $N = ?$

$$p_2/p_1 = \varepsilon^N \quad \lg(p_2/p_1) = N \cdot \lg \varepsilon$$