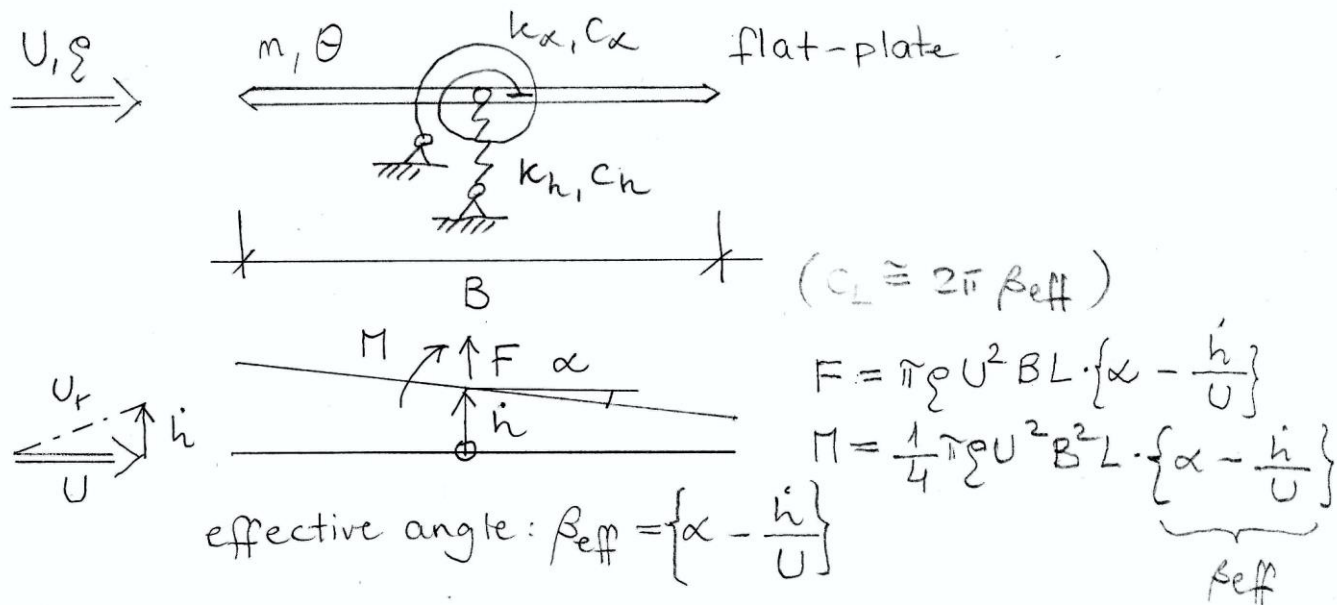


Flutter I.

load models: - quasi-steady
- unsteady

quasi-steady approach:



$$\begin{bmatrix} m & \\ & \theta \end{bmatrix} \begin{bmatrix} \ddot{h} \\ \ddot{\alpha} \end{bmatrix} + \begin{bmatrix} c_h & \\ & c_\alpha \end{bmatrix} \begin{bmatrix} \dot{h} \\ \dot{\alpha} \end{bmatrix} + \begin{bmatrix} k_h & \\ & k_\alpha \end{bmatrix} \begin{bmatrix} h \\ \alpha \end{bmatrix} = \underline{Q}(x, \dot{x}) \quad x = \begin{bmatrix} h \\ \alpha \end{bmatrix}$$

$$= \pi \rho U^2 B L \cdot \left\{ \begin{array}{l} \alpha - \dot{h}/U \\ B/4 (\alpha - \dot{h}/U) \end{array} \right\}$$

$$\begin{bmatrix} m & \\ & \theta \end{bmatrix} \begin{bmatrix} \ddot{h} \\ \ddot{\alpha} \end{bmatrix} + \begin{bmatrix} c_h + \pi \rho U B L & 0 \\ \frac{1}{4} \pi \rho U B^2 L & c_\alpha \end{bmatrix} \begin{bmatrix} \dot{h} \\ \dot{\alpha} \end{bmatrix} + \begin{bmatrix} k_h - \pi \rho U^2 B L & \\ 0 & k_\alpha - \frac{1}{4} \pi \rho U^2 B^2 L \end{bmatrix} \begin{bmatrix} h \\ \alpha \end{bmatrix} = \underline{0}$$

$$\underline{M} \ddot{x} + \underline{C} \dot{x} + \underline{K} x = \underline{0} \quad x = \underline{\tilde{\phi}} \cdot e^{i\omega t}$$

$$(\underline{C}_S + \underline{C}_{AE})$$

$$(x = \underline{\tilde{\phi}} \cdot e^{\gamma t} \text{ alternatively})$$

$$\left[-\omega^2 \underline{M} + i\omega \underline{C} + \underline{K} \right] \underline{\tilde{\phi}} = \underline{0}$$

$$\tilde{\omega} = a + ib$$

$$e^{i\tilde{\omega}t} = e^{iat} \cdot e^{i(ib)t} = e^{i\omega t} \cdot e^{-\gamma t}$$

$$\rho(U) = ?$$

$$\rho > 0, \text{ stab.}$$

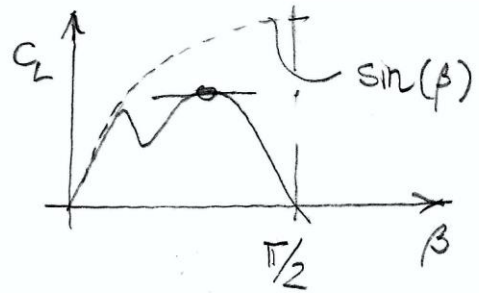
$$\rho < 0, \text{ instab.}$$

$$\rho = 0, \text{ flutter onset}$$

Hopf bifurcation

$$\underline{M} \ddot{\underline{x}} + \underline{C} \dot{\underline{x}} + \underline{K} \underline{x} = \underline{Q}(\underline{x}, \dot{\underline{x}}, U)$$

$$\underline{Q} = \begin{cases} \pi \rho U^2 B L \cdot \sin(\alpha - h/U) \\ 1/4 \pi \rho U^2 B^2 L \cdot \sin(\alpha - h/U) \end{cases}$$



$$\sin(\alpha - h/U) \approx \left\{ \alpha - h/U \right\} + \left\{ -\frac{1}{6} \alpha^3 + \frac{1}{2U} \alpha^2 h^2 - \frac{1}{2U^2} \alpha h^2 + \frac{1}{6U^3} h^3 \right\}$$

$$\underline{x} = \begin{bmatrix} h \\ \alpha \end{bmatrix}; \quad \dot{\underline{x}} = \begin{bmatrix} \dot{h} \\ \dot{\alpha} \end{bmatrix}; \quad \ddot{\underline{x}} = \begin{bmatrix} \ddot{h} \\ \ddot{\alpha} \end{bmatrix}$$

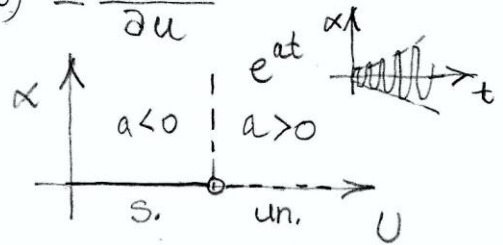
↓ Cauchy

$$\dot{\underline{z}} = \underline{A} \underline{z} + \underline{g}(\underline{z})$$

$$z(1) = \alpha$$

$$z(2) = h$$

$$\sigma_0'(0) = \frac{\partial a(U)}{\partial U}$$



$$\left[\underline{A}(U) - \lambda \underline{I} \right] \tilde{\underline{\phi}} = \underline{0} \rightarrow \lambda = a + ib$$

↓ Hopf

$$\underline{T} = \begin{bmatrix} \text{Re}(\tilde{\underline{\phi}}_1) & \text{Im}(\tilde{\underline{\phi}}_2) & \text{Re}(\tilde{\underline{\phi}}_2) & \text{Im}(\tilde{\underline{\phi}}_1) \end{bmatrix}; \quad \underline{z} = \begin{bmatrix} h \\ h \\ \alpha \\ \alpha \end{bmatrix}; \quad \tilde{\underline{\phi}} = \begin{bmatrix} 1 \\ 1 \\ 1 \\ 1 \end{bmatrix}$$

$$\underline{z} = \underline{T} \underline{u} \quad (\sim \text{modal analysis})$$

$$\underline{T} \dot{\underline{u}} = \underline{A} \underline{T} \underline{u} + \underline{g}(\underline{u}) / \underline{T}^{-1} \rightarrow$$

$$\dot{\underline{u}} = \underbrace{\underline{T}^{-1} \underline{A} \underline{T}}_{\underline{A}} \underline{u} + \underbrace{\underline{T}^{-1} \cdot \underline{g}(\underline{u})}_{\underline{b}}$$

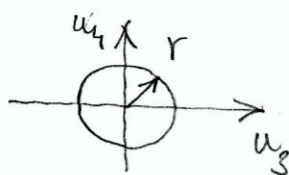
$$\underline{A} = \begin{bmatrix} \sigma_1 & \omega_1 & 0 & 0 \\ -\omega_1 & \sigma_2 & 0 & 0 \\ 0 & 0 & \omega_2 & 0 \\ 0 & \omega_2 & 0 & 0 \end{bmatrix}$$

$$\underline{b} = \begin{bmatrix} - \\ - \\ \sum_{ij} a_{ij} \cdot u_3^i \cdot u_4^j \\ \sum_{ij} b_{ij} \cdot u_3^i \cdot u_4^j \end{bmatrix}$$

$$\delta(0) = \frac{1}{8} \cdot \left\{ 3a_{03} + a_{12} + b_{21} + 3b_{03} \right\} + \frac{1}{\omega} \cdot \left\{ \dots \right\}$$

$\delta(0) < 0$: supercritical
 $\delta(0) > 0$: subcritical

↓ amplitudes



$$\alpha(U) = r(U) \cdot |\tilde{\underline{\phi}}_2(3)|$$

$$r = \sqrt{\frac{-\sigma_0'(0)}{\delta(0)} \cdot [U - U_{cr}]}$$

