

# Advanced Fluid Mechanics

BME GEÁT MW01

Dr. Gergely Kristóf  
 Department of Fluid Mechanics, BME  
 February 2019.

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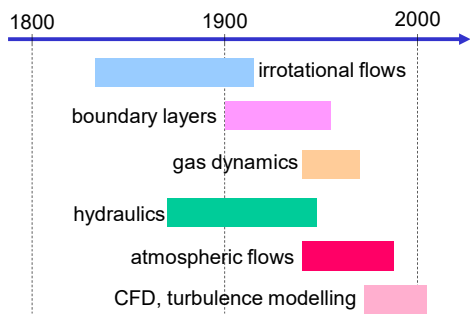
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## Timeline




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## Foreseeable program

Course week	Topics
1.	Overview of the fundamentals of fluid mechanics. Vorticity transport equation.
2.	Irrrotational flows, Darcy flow. Wells.
3.	Joukowski transformation.
4.	Boundary layers. Similarity solutions for laminar and turbulent boundary layers.
5.	Transition. Turbulent boundary layers. BL control.
6.	Fundamentals of gasdynamics. Wave phenomena. Izentropic flow
7.	Normal shock waves
8.	Oblique shock waves, wave reflection. Prandtl-Meyer expansion, Supersonic jets.
9.	Pipe networks
10.	Pipe transients
11.	Atmospheric flows
12.	--- (Faculty Day of Profession)
13.	--- (May 1)
14.	Martijn Schoot: Application of computational fluid dynamics (CFD) in pump design.

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## References

**Lecture handouts:**

[http://www.ara.bme.hu/oktatas/tantargy/NEPTUN/BMEGEATMW01/2018-2019-II/ea\\_lecture/](http://www.ara.bme.hu/oktatas/tantargy/NEPTUN/BMEGEATMW01/2018-2019-II/ea_lecture/)

**For further reading:**

- 1) Lamb H: Hydrodynamics, 1932.
- 2) Schlichting H: Boundary Layer Theory, 1955.
- 3) Shapiro A. H: The Dynamics and Thermodynamics of Compressible Fluid Flow, 1953.
- 4) Streeter V. L, Wylie E. B: Fluid Mechanics, McGraw-Hill, 1975.
- 5) Ferziger J. H, Peric M: Computational Methods for Fluid Dynamics, Springer, ISBN 3-540-42074-6, 2002.

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## 1. Vorticity transport

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## Acceleration of a fluid parcel

Motion of clouds: [rudder](#), [satellite](#)

Velocity components:  $\vec{v}(t, \vec{r}) = u(t, x, y, z)\vec{i} + v(t, x, y, z)\vec{j} + w(t, x, y, z)\vec{k}$

$$du = \frac{\partial u}{\partial t} dt + \frac{\partial u}{\partial x} dx + \frac{\partial u}{\partial y} dy + \frac{\partial u}{\partial z} dz$$

For a fluid parcel:  $\frac{dx}{dt} = u, \frac{dy}{dt} = v, \frac{dz}{dt} = w$

Substantial derivatives	$\left. \begin{aligned} \frac{du}{dt} &= \frac{\partial u}{\partial t} + \vec{v} \cdot \nabla u \\ \frac{dv}{dt} &= \frac{\partial v}{\partial t} + \vec{v} \cdot \nabla v \\ \frac{dw}{dt} &= \frac{\partial w}{\partial t} + \vec{v} \cdot \nabla w \end{aligned} \right\}$	$\frac{d\vec{v}}{dt} = \frac{\partial \vec{v}}{\partial t} + \vec{v} \cdot \nabla \vec{v}$
		velocity gradient tensor $\swarrow \quad \searrow$ local acceleration      convective acceleration

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## Navier-Stokes equation

$$\rho = \text{const}, \quad \nu = \text{const}$$

$$\frac{d\vec{v}}{dt} = \frac{\partial \vec{v}}{\partial t} + \vec{v} \cdot \nabla \vec{v} = -\frac{1}{\rho} \nabla p + \vec{g} + \nu \Delta \vec{v} \quad \left[ \frac{N}{kg} \right]$$

pressure force
body-force
shear

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## Continuity

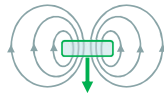
$$\frac{\partial}{\partial t} \int_V \rho dV + \oint_A \rho \vec{v} \cdot d\vec{A} = 0 \quad \text{which yields:} \quad \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{v}) = 0$$

accumulated mass
mass outflux

If  $\rho = \text{const}$ :  $\nabla \cdot \vec{v} = 0$  thus  $\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0$

Stream tubes can form closed tubes or have both ends on boundaries.

Eg. absolute streamlines around a moving object:




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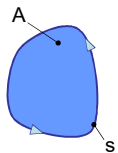
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## Vortices



Circulation:

$$\Gamma = \oint_s \vec{v} \cdot d\vec{s}$$

Vorticity:

$$\vec{\omega} = \nabla \times \vec{v} = \begin{pmatrix} w_y - v_z \\ u_z - w_x \\ v_x - u_y \end{pmatrix}$$

From the Stokes theorem: On a singly-connected surface.  $\Gamma = \int_A \vec{\omega} \cdot d\vec{A}$

Can we have circulation without vorticity?

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### Free vortex

Vorticity is 2 times the angular velocity of the fluid parcel.

You must exclude this central part, otherwise then the domain is singly connected and circulation is not possible.

Vorticity is concentrated in the center.

$\omega_z = 0$

$\Gamma_1 = -\Gamma_2$

$2r\pi v = \text{const.}$

$v = \frac{\text{const.}}{r}$

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### Thomson theorem

If the the integration path  $S$  is a fluid line of a perfect fluid, then  $\frac{d\Gamma}{dt} = 0$

Kutta-Joukowski theorem:  $F_{\text{lift}} = \rho v_{\infty} \Gamma$

Circulation is proportional with the lift force.

How vorticity is produced?

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### Evolution of vorticity

Vorticity transport equation:  $\nabla \times (\text{Navier - Stokes})$

Let's derive it in 2D!  $\omega$  is a scalar in 2D:  $\omega = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$

By taking the curl of N-S equation:

$$\frac{\partial}{\partial x} \left( \frac{\partial v}{\partial t} \right) + \frac{\partial}{\partial x} \left( u \frac{\partial v}{\partial x} \right) + \frac{\partial}{\partial x} \left( v \frac{\partial v}{\partial y} \right) = -\frac{1}{\rho} \frac{\partial}{\partial x} \left( \frac{\partial p}{\partial y} \right) + \frac{\partial g_y}{\partial x} + \nu \frac{\partial}{\partial x} \left( \frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right)$$

$$\frac{\partial}{\partial y} \left( \frac{\partial u}{\partial t} \right) + \frac{\partial}{\partial y} \left( u \frac{\partial u}{\partial x} \right) + \frac{\partial}{\partial y} \left( v \frac{\partial u}{\partial y} \right) = -\frac{1}{\rho} \frac{\partial}{\partial y} \left( \frac{\partial p}{\partial x} \right) + \frac{\partial g_x}{\partial y} + \nu \frac{\partial}{\partial y} \left( \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$


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$$\frac{\partial \omega}{\partial t} + u \frac{\partial \omega}{\partial x} + v \frac{\partial \omega}{\partial y} + *** = 0 + 0 + \nu \left( \frac{\partial^2 \omega}{\partial x^2} + \frac{\partial^2 \omega}{\partial y^2} \right)$$

$$*** = \frac{\partial u}{\partial x} \frac{\partial v}{\partial x} + \frac{\partial v}{\partial x} \frac{\partial v}{\partial y} - \frac{\partial u}{\partial y} \frac{\partial u}{\partial x} - \frac{\partial v}{\partial y} \frac{\partial u}{\partial y} = \frac{\partial v}{\partial x} \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) - \frac{\partial u}{\partial y} \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right)$$

Because of the continuity:  $0 = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y}$      $0 = \frac{\partial u}{\partial y} - \frac{\partial v}{\partial x}$

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### Curl of the convective acceleration term in 3D flow

$$\nabla \times (\vec{v} \cdot \nabla \vec{v}) = \nabla \times \begin{pmatrix} u_x & u_y & u_z \\ v_x & v_y & v_z \\ w_x & w_y & w_z \end{pmatrix} \begin{pmatrix} u \\ v \\ w \end{pmatrix} = \vec{\omega} = \nabla \times \begin{pmatrix} u \\ v \\ w \end{pmatrix} = \begin{pmatrix} w_y - v_z \\ u_z - w_x \\ v_x - u_y \end{pmatrix}$$

$$= \begin{pmatrix} w_{xy} - v_{xz} & w_{yy} - v_{yz} & w_{zy} - v_{zz} \\ u_{xz} - w_{xx} & u_{yz} - w_{yx} & u_{zz} - w_{zx} \\ v_{xx} - u_{xy} & v_{yx} - u_{yy} & v_{zz} - u_{zy} \end{pmatrix} \begin{pmatrix} u \\ v \\ w \end{pmatrix} + \begin{pmatrix} w_x u_y - v_x u_z + w_y v_y - v_y v_z + w_z w_y - v_z w_z \\ u_x u_z - w_x u_x + u_y v_z - w_y v_x + u_z w_z - w_z w_x \\ v_x u_x - u_x u_y + v_y v_x - u_y v_y + v_z w_x - u_z w_y \end{pmatrix}$$

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$$\nabla \times (\vec{v} \cdot \nabla \vec{v}) = \begin{pmatrix} \alpha \\ \beta \\ \gamma \end{pmatrix} = \begin{pmatrix} w_y - v_z \\ u_z - w_x \\ v_x - u_y \end{pmatrix}$$

$$\begin{pmatrix} \alpha_x & \alpha_y & \alpha_z \\ \beta_x & \beta_y & \beta_z \\ \gamma_x & \gamma_y & \gamma_z \end{pmatrix} \begin{pmatrix} u \\ v \\ w \end{pmatrix} + \begin{pmatrix} w_x u_y - v_x u_z - u_x w_y + u_x v_z \\ u_y v_z - w_y v_x - v_y u_z + v_y w_x \\ v_z w_x - u_z w_y - w_z v_x + w_z u_y \end{pmatrix} + \begin{pmatrix} \alpha \\ \beta \\ \gamma \end{pmatrix} (u_x + v_y + w_z)$$

$$\vec{\omega} \cdot \nabla \vec{v} = \begin{pmatrix} u_x w_y - u_x v_z + u_y u_z - u_y w_x + u_z v_x - u_z u_y \\ v_x w_y - v_x v_z + v_y u_z - v_y w_x + v_z v_x - v_z u_y \\ w_x w_y - w_x v_z + w_y u_z - w_y w_x + w_z v_x - w_z u_y \end{pmatrix}$$

$$\nabla \times (\vec{v} \cdot \nabla \vec{v}) = \vec{v} \cdot \nabla \vec{\omega} - \vec{\omega} \cdot \nabla \vec{v} + \vec{\omega} \nabla \cdot \vec{v}$$

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### Vorticity transport

(ρ and v are constants)

$$\frac{d\vec{v}}{dt} = \frac{\partial \vec{v}}{\partial t} + \vec{v} \cdot \nabla \vec{v} = -\frac{1}{\rho} \nabla p + \vec{g} + \nu \Delta \vec{v} \quad \nabla \times \dots$$

$$\frac{d\vec{\omega}}{dt} = \frac{\partial \vec{\omega}}{\partial t} + \vec{v} \cdot \nabla \vec{\omega} = 0 + \nabla \times \vec{g} + \nu \Delta \vec{\omega} - \vec{\omega} \nabla \cdot \vec{v} + \vec{\omega} \cdot \nabla \vec{v}$$

vortex advection
0, if g is irrotational
vortex diffusion
0
vortex stretching

$$\frac{d\vec{\omega}}{dt} = \nu \Delta \vec{\omega} + \vec{\omega} \cdot \nabla \vec{v}$$

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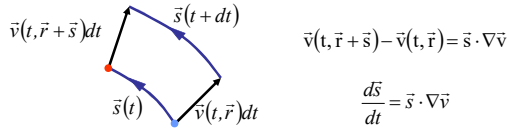
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## What is vortex stretching?

Evolution of a fluid line of elementary length



Vorticity transport equation for irrotational body force and zero viscosity:

$$\frac{d\vec{\omega}}{dt} = \vec{\omega} \cdot \nabla \vec{v}$$

The direction of  $\underline{s}$  is arbitrarily chosen.

Both vectors evolve according to the same transport equation, hence, in inviscid flow, the vorticity vector behaves in the same way as an infinitesimal fluid line element. (Helmholtz)

Thus,  $\underline{\omega}$  will grow, when the fluid line is stretched.

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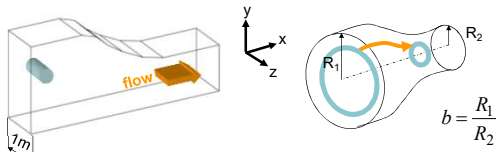
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## Problem #1.1



Compare a 2D confuser (of slab symmetry) with an axial symmetric confuser:

- What components of the vorticity vector are non-zero? Use cylindrical coordinates  $(x, r, \phi)$  in the axisymmetric case!
- In what proportion does the length of a fluid element change?
- In what proportion will the components of the vorticity change if the vortex diffusion is negligible?

To the solution

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## Vortex diffusion

The vorticity transport equation for a 2D flow of a constant property Newtonian fluid:

$$\frac{d\omega}{dt} = \frac{\partial \omega}{\partial t} + \vec{v} \cdot \nabla \omega = \nu \Delta \omega \quad \nu = \frac{\mu}{\rho} \left[ \frac{m^2}{s} \right]$$

kinematical viscosity

Is in full analogy with the heat transport equation:

$$\frac{dT}{dt} = \frac{\partial T}{\partial t} + \vec{v} \cdot \nabla T = a \Delta T \quad a = \frac{\lambda}{\rho c} \left[ \frac{m^2}{s} \right]$$

heat diffusion coefficient

The kinematical viscosity can be regarded as a vorticity diffusion coefficient. These two phenomena are in full analogy.

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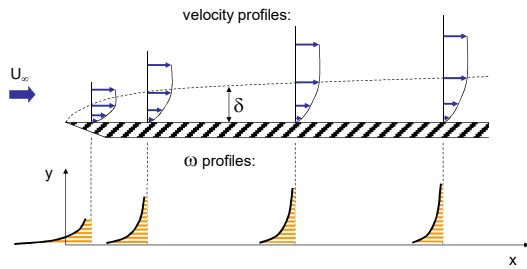
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## Boundary layer over a flat plate



Vorticity is continuously produced on the wall surface due to the no-slip condition, and it is conducted into the main stream by the viscosity.

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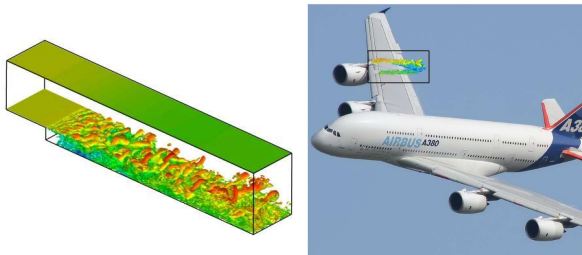
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## The role of advection



[http://www.computationalfluidynamics.com.au/cfd-turbulence-part5-scale-resolving-simulations\\_srs/](http://www.computationalfluidynamics.com.au/cfd-turbulence-part5-scale-resolving-simulations_srs/)

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## Summary

The vorticity transport equation for incompressible fluids reads:

$$\frac{d\vec{\omega}}{dt} = \nabla \times \vec{g} + \nu \Delta \vec{\omega} + \vec{\omega} \cdot \nabla \vec{v}$$

Origin of vorticity:

- Boundary conditions (wall shear)
- Non conservative forces (eg. Coriolis force)

Redistribution of vorticity:

- Advection
- Vortex stretching
- Vortex diffusion

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Solution

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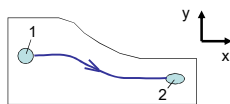
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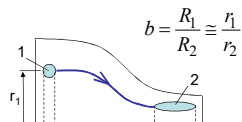
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Solution of problem #1.1



$\omega_x = 0$     $\omega_y = 0$     $\omega_z \neq 0$

$s_{z,2} = s_{z,1} \longrightarrow \omega_{z,2} = \omega_{z,1}$



$\omega_x \neq 0$     $\omega_r = 0$     $\omega_\phi \neq 0$

$s_{\phi,2} = s_{\phi,1} b^{-1} \longrightarrow \omega_{\phi,2} = \omega_{\phi,1} b^{-1}$

from the velocity magnification:

$s_{x,2} = s_{x,1} b^2 \longrightarrow \omega_{x,2} = \omega_{x,1} b^2$

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